

# Wave Forces Acting on a Blunt Ship Advancing in Oblique Short Waves

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## Introduction

Results of numerical and experimental study on the wave forces acting on a blunt ship advancing in oblique short waves are presented in this paper. We have been continuously investigated about this theme in these three years, and already published two papers as a result [1,2]. The main purpose of that study was to investigate the efficiencies of the 3-D panel method for predicting wave forces acting on a blunt ship especially in oblique short waves, and to make the 3-D panel method robust and practical scheme for that prediction. In order to achieve the purpose, a lot of numerical calculations and experiments were performed not only for the wave forces such as wave exciting forces and second order steady forces but also for the wave pressure on the ship surface. These plenty of calculations, experiments and detail comparison between them confirmed that 3-D panel method is generally useful to predict the wave forces comparatively in long waves, while in short waves it does not give such accurate solutions as be in good agreement with experiments even if the large number of panels is used.

Many calculations due to 3-D panel method have been presented by many researchers up to now [3,4,5], and another new predicting schemes such as Rankine panel method have been also developed recently [6,7,8]. Most of those numerical results, although, are for the slender ship or for the long wave case unfortunately. It will be difficult to confirm that enough investigations as to the predicting scheme of wave forces acting on the blunt ship advancing in short waves have been done, or that robust and practical method for that purpose has been established already.

We are now performing the new numerical calculations based on the newly formulated 3-D panel method, and those results will be presented with detail experimental results in this paper. The aforementioned discrepancies between calculations and experiments in oblique short waves will be solved by this new calculations.

## Formulation

We consider a ship advancing at constant forward speed  $U$  in regular waves encountering with angle  $\chi$ , and take the coordinate system illustrated in Fig.1. The circular frequency, wave number and wave amplitude of incident wave are  $\omega_0, K$  and  $A$  respectively, and encounter circular frequency is  $\omega_e (= \omega_0 - KU \cos \chi)$ . The linear theory is employed for this problem assuming incompressible, inviscid and irrotational fluid.

The velocity potential  $\Phi$  governed by Laplace's equation can be expressed as

$$\Phi(x, y, z; t) = U[-x + \phi_s(x, y, z)] + Re[\phi(x, y, z)e^{i\omega_e t}] \quad (1)$$

where

$$\phi = \frac{gA}{\omega_0}(\phi_0 + \phi_7) + i\omega_e \sum_{j=1}^6 \xi_j \phi_j \quad (2)$$

$$\phi_0 = ie^{Kz - iK(x \cos \chi + y \sin \chi)} \quad (3)$$

$\phi_s$  is the steady velocity potential independent of time, and  $\phi$  is the unsteady velocity potential including the incident wave potential  $\phi_0$ , the radiation potentials  $\phi_j$  ( $j = 1 \sim 6$ ) and the diffraction potential  $\phi_7$ .  $\xi_j$  indicates the complex amplitude of ship motion in  $j$ -th direction.

The unsteady disturbance potential  $\phi_j$  ( $j = 1 \sim 7$ ) must supply the following free-surface boundary condition and body boundary condition.

$$\left(i\omega_e - U \frac{\partial}{\partial x}\right)^2 \phi_j + \mu \left(i\omega_e - U \frac{\partial}{\partial x}\right) \phi_j + g \frac{\partial \phi_j}{\partial z} = 0 \quad z = 0 \quad (4)$$

$$\frac{\partial \phi_j}{\partial n} = n_j + \frac{U}{i\omega_e} m_j \quad (j = 1 \sim 6) \quad , \quad \frac{\partial \phi_7}{\partial n} = -\frac{\partial \phi_0}{\partial n} \quad \text{on } S_H \quad (5)$$

where

$$\begin{aligned} (n_1, n_2, n_3) &= \mathbf{n} ; \text{ inward normal} \quad , \quad (n_4, n_5, n_6) = \mathbf{r} \times \mathbf{n} \quad , \\ (m_1, m_2, m_3) &= -(\mathbf{n} \cdot \nabla) \mathbf{V} \quad , \quad (m_4, m_5, m_6) = -(\mathbf{n} \cdot \nabla)(\mathbf{r} \times \mathbf{V}) \quad , \\ \mathbf{V} &= \nabla(-x + \phi_s) \quad , \quad \mathbf{r} = (x, y, z) \end{aligned}$$

$m_j$  in eq.(5), which shows the effect of the steady flow field on the ship surface, tends to be estimated approximately assuming that the free surface is not disturbed by the ship's forward translation.

Application of the Green's second identity for the closed region surrounded by  $S_H, S_F$  and artificial surface located at the infinity leads to the integral expression of  $\phi_j$  as follows:

$$\begin{aligned} \phi_j(P) &= - \iint_{S_H} \left[ \frac{\partial \phi_j(Q)}{\partial n} - \phi_j(Q) \frac{\partial}{\partial n} \right] G(P, Q) dS \\ &\quad + \frac{1}{K_0} \oint_{C_H} \left[ \frac{\partial \phi_j(Q)}{\partial x'} - \phi_j(Q) \frac{\partial}{\partial x'} - 2i\tau K_0 \phi_j(Q) \right] G(P, Q) dy' \end{aligned} \quad (6)$$

where  $P \equiv (x, y, z), Q \equiv (x', y', z')$  show the field point and source point respectively.  $G(P, Q)$  denotes the Green function which satisfies the free-surface condition (4) and the radiation condition at infinity, and can be evaluated accurately by means of Ohkusu & Iwashita's method [9,10].

In order to solve the integral equation derived from eq.(6), we express the body shape and the potential distribution on it by means of the spline function. After discretizing the body surface into  $N$  elements, the arbitrary point on it can be expressed as

$$x(\xi, \eta) = \sum_{j=1}^N \alpha_j B_j(\xi, \eta), \quad y(\xi, \eta) = \sum_{j=1}^N \beta_j B_j(\xi, \eta), \quad z(\xi, \eta) = \sum_{j=1}^N \gamma_j B_j(\xi, \eta) \quad (7)$$

where  $\xi$  and  $\eta$  are the local coordinates situated on the body surface, and  $B_j(\xi, \eta)$  is the two dimensional spline function. The spline coefficients  $\alpha_j, \beta_j$  and  $\gamma_j$  are determined from the coordinates of the collocation points on  $N$  elements. The transformation of the velocity fields between global and local coordinate systems are

$$\begin{bmatrix} \partial \phi_j / \partial x \\ \partial \phi_j / \partial y \\ \partial \phi_j / \partial z \end{bmatrix} = \begin{bmatrix} \partial \mathbf{r} / \partial \xi \\ \partial \mathbf{r} / \partial \eta \\ \partial \mathbf{r} / \partial \zeta \end{bmatrix}^{-1} \begin{bmatrix} \partial \phi_j / \partial \xi \\ \partial \phi_j / \partial \eta \\ \partial \phi_j / \partial \zeta \end{bmatrix} \quad (8)$$

where  $\zeta$ -axis is taken to be vertical to both  $\xi$  and  $\eta$  axes, and the partial derivatives of  $x, y$  and  $z$  with respect to  $\zeta$  are evaluated by

$$\frac{\partial \mathbf{r}}{\partial \zeta} = \frac{\partial \mathbf{r}}{\partial \xi} \times \frac{\partial \mathbf{r}}{\partial \eta}$$

The similar expression as the geometrical expression of the body shape shown in eq.(7) is employed also for the potential distribution on the body surface such as

$$\phi_j(\xi, \eta) = \sum_{i=1}^N \lambda_i B_i(\xi, \eta) \quad (9)$$

$\lambda_j$  is the unknown spline coefficients. By using this expression of the potential distribution, we can express its derivatives in terms of the derivatives of the spline function.

The discretized integral equation to be solved follows from eq.(6) and (9) in the form

$$\begin{aligned} & \sum_{j=1}^N \left[ \sum_{l=1}^N \left( -\hat{B}(l, j) \iint_{\Delta S(l)} \hat{G}_n(k, l) dS \right. \right. \\ & \quad - \frac{1}{K_0} \left\{ [C_{11}(l)\hat{B}_\xi(l, j) + C_{12}(l)\hat{B}_\eta(l, j)] \oint_{\Delta C(l)} \hat{G}(k, l) dy' \right. \\ & \quad \left. \left. - \hat{B}(l, j) \oint_{\Delta C(l)} [\hat{G}_{x'}(k, l) + 2iK_0\tau\hat{G}(k, l)] dy' \right\} \right] \lambda_j \\ & = - \sum_{l=1}^N \phi_{jn}(l) \left[ \iint_{\Delta S(l)} \hat{G}(k, l) dS - \frac{1}{K_0} C_{13}(l) \oint_{\Delta C(l)} \hat{G}(k, l) dy' \right] \quad (k = 1 \sim N) \quad (10) \end{aligned}$$

$\hat{B}(i, j)$  indicates  $B_j(\xi_i, \eta_i)$ , and  $\hat{B}_\xi(i, j), \hat{B}_\eta(i, j)$  are its partial derivatives with respect to  $\xi$  and  $\eta$  respectively.  $\hat{G}(k, l)$  and  $\hat{G}_n(k, l)$  are the Green function and its normal derivative, which are evaluated at the  $k$ -th collocation point as the contribution from  $l$ -th element.  $C_{11}(l), C_{12}(l)$  and  $C_{13}(l)$  denote the elements of the matrix in eq.(8). In this formulation,  $\phi_j$  and  $\partial\phi_j/\partial x$  are assumed to be constants over the elements which touch with the free surface, and the calculation of the kernel for each elements is carried out following the manner of the constant panel. Eq.(10) is solved numerically with respect to the unknown spline coefficients  $\lambda_j$  submitting boundary conditions into  $\phi_{jn}(l)$ , that is, the normal derivative of the velocity potential.

Once the coefficients  $\lambda_j$  have been determined, the velocity potential and velocity fields may be easily calculated by means of eq.(9). The calculations for other physical values such as linearized unsteady pressure and wave forces are straightforward following well known formulations based on the linear theory.

## Numerical Results

Two kinds of calculations are performed and compared. One is the calculation based on the formulation presented in this paper. Another one is the calculation based on the formulation due to the source distribution, which is well known and alternative expression of the velocity potential (6). For the calculations based on the later formulation, no interpolating function such as spline function related above is used and only the manner of the constant panel is applied.

Fig.2 and 4 show the panel distribution of the blunt ship ( $L/B = 5$ ) and calculated wave exciting force for surge respectively, which have been already reported in [1]. Calculations are carried out following the formulation due to the source distribution. The large discrepancies are observed between numerical and experimental results for the case with forward speed.

Fig.5 is the present results calculated for the half-submerged prolate spheroid ( $L/B = 5$ ) illustrated in Fig.3. It is confirmed that the present formulation gives reasonable results compared with another formulation based on the source distribution. The differences between two formulations may be caused by such phenomenon as the irregular frequency which is possible for the formulation due to the source distribution. The similar reasonable results can be expected even for the real ship in Fig.2. We are now trying to apply the present scheme to the real blunt ship, and several results will be appear until the workshop.

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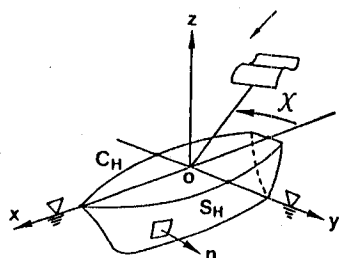


Fig.1 Coordinate system

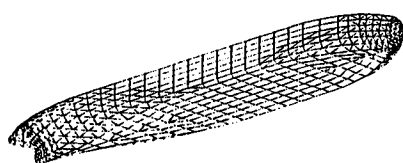


Fig.2 Panel distribution on the blunt ship  
( $N = 994$ )

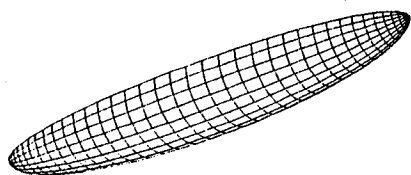


Fig.3 Panel distribution on the half-submerged prolate spheroid ( $N = 480$ )

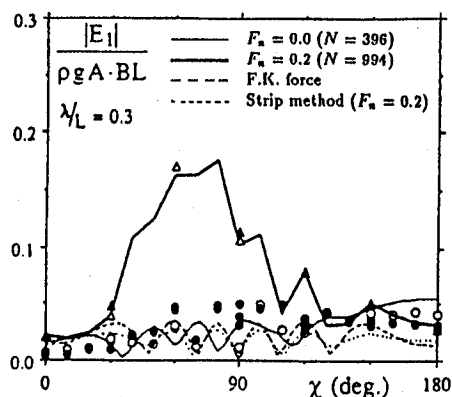


Fig.4 Wave exciting force for surge in oblique short waves ( $\lambda/L = 0.3$ , blunt ship)

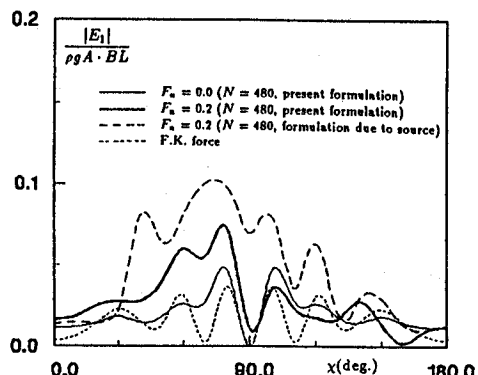


Fig.5 Wave exciting force for surge in oblique short waves ( $\lambda/L = 0.3$ , half-submerged prolate spheroid)

## DISCUSSION

**Newman J.N.:** 1) What kind of spline basis functions did you use?

2) How did you evaluate the integrals over each panel?

**Iwashita H.:** 1) Cubic B-spline is used. Although another kind of B-spline such as 5-th order B-spline was also tried, we could not find out any advantages comparing Cubic B-spline.

2) The following scheme is used for evaluating

$$J \equiv \int_{a_1}^{b_1} \int_{a_2}^{b_2} f(x_1, x_2) dx_1 dx_2$$

- for the normal function  $f(x_1, x_2)$

Clenshaw-Curtis method

- for the function  $f(x_1, x_2)$  which has the singularity

Integral transformation + Clenshaw-Curtis method + shift of the field point of  $O(10^{-10})$

$$J = \frac{(b_1 - a_1)(b_2 - a_2)}{4} \times \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} f(x_1, x_2) \frac{1.5 \cosh t_1}{\cosh^2(1.5 \sinh t_1)} \frac{1.5 \cosh t_2}{\cosh^2(1.5 \sinh t_2)} dt_1 dt_2$$

$$\left. \begin{aligned} x_1 &= [(b_1 - a_1) \tanh(1.5 \sinh t_1) + (b_1 + a_1)]/2 \\ x_2 &= [(b_2 - a_2) \tanh(1.5 \sinh t_2) + (b_2 + a_2)]/2 \end{aligned} \right\}$$

**Kim M.H.:** According to your numerical results, there is big discrepancy between the constant panel result and the high-order element result. Is it a generic numerical problem of the constant panel method for this particular application? Or can you have better agreement by simply increasing the number of constant panels?

**Iwashita H.:** The discrepancy between upper part and lower part of the figure which I presented doesn't have any relation with the approximation order of the element at all. This discrepancy is caused by the different type of the integral equation. If we solve the integral equation with respect to  $\phi$  assuming constant panel (0-th order) approximation, we can get a similar solution to that obtained by the assumption of higher-order approximation, provided the number of panel is enough (at least more than 100 elements).